



06/02/09

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WASHINGTON, D C 20231

Case Docket No. 1759/2

Sir:

Transmitted herewith for filing is the patent application of

Inventor: JOSEPH NIR, BARUCH SHAYEVICH AND HANOCH COHENFor : COHERENT PROCESSING OF SATELLITE SIGNALS TO LOCATE A MOBILE UNIT

Enclosed are:

- ☒ 4 sheets of informal drawing(s)
- ☒ An assignment of the invention to CELLGUIDE LTD.
- ☐ A certified copy of a \_\_\_\_\_ application
- ☐ An associate power of attorney.
- ☒ A verified statement to establish small entity status under 37 CFR 1.9 and 37 CFR 1.27.
- ☐ Other - \_\_\_\_\_

The filing fee has been calculated as shown below.

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FOR	NO. FILED	NO. EXTRA	
BASIC FEE			
TOTAL CLAIMS	38 - 20 =	18	
INDEP CLAIMS	6 - 3 =	3	
Record of Assignment		40	

If the difference in Col.1 is less  
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SMALL ENTITY	
RATE	FEE
	\$ 34
18 x9 =	\$ 162
3 x39	\$ 117
	\$ 40
TOTAL	\$ 353

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RATE	FEE
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x18 =	\$
x78	\$
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- ☒ Any filing fees under 37 CFR 1.16 for presentation of extra claims.

Respectfully,

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06/02/09

SMALL BUSINESS CONCERN - NEW APPLICATION

Attorney Docket No.: 1759/2

IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

In RE Application of: JOSEPH NIR, BARUCH SHAYEVICH AND HANCOH COHEN

Filed Concurrently Herewith

For: COHERENT PROCESSING OF SATELLITE SIGNALS TO LOCATE A MOBILE UNIT

VERIFIED STATEMENT UNDER 37 CFR 1.27  
CLAIMING STATUS AS A SMALL ENTITY

To The Commissioner of Patents and Trademarks:

I hereby declare that:

I am the owner of, or an official empowered to act on behalf of, the small business concern identified below:

Name of Concern: CELLGUIDE LTD.

Address : BERGMAN 2, 76705 REHOVOT, ISRAEL

The small business concern identified above, together with its affiliates, employs fewer than 500 persons and qualifies as a small business concern as defined in 37 CFR 1.9(d) for purposes of paying reduced fees under 35 USC § 41(a) and § 41(b) to the Patent and Trademark Office with regard to the above-entitled invention described in the specification filed herewith.

Rights under contract or law have been conveyed to and remain with the small business concern identified above with regard to the above entitled invention.

If the rights held by the small business concern are not exclusive, each other party having rights to the invention is listed below, and no rights to the invention are held by any party who could not qualify as a small entity under 37 CFR 1.9(f), namely any person who could not be classified as an independent inventor under 37 CFR 1.9(c) if that person had made the invention, or any concern which would not qualify as a small business concern under 37 CFR 1.9(d) or a nonprofit organization under 37 CFR 1.9(e).

Full Name (Party 1) : NONE

Address : \_\_\_\_\_

Status : ☐ Individual

☐ Small Business  
Concern

☐ Nonprofit  
Organization

Full Name (Party 2) : \_\_\_\_\_

Address : \_\_\_\_\_

Status : ☐ Individual

☐ Small Business  
Concern

☐ Nonprofit  
Organization

I acknowledge the duty under 37 CFR 1.28(b) to file, in this application, notification of any change in status resulting in loss of entitlement to small entity status prior to paying, or at the time of paying, the issue fee due after the date on which status as a small entity is no longer appropriate.

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application and any patent issuing thereon.

NIR JOSEPH  
Name of Person Signing

NIR JOSEPH  
Signature

12/26/99  
Date

Capacity of Person Signing: General Manager Director

Address of Person Signing : \_\_\_\_\_

JC933 U.S. PTO  
09/586619  
06/02/00

INDEPENDENT INVENTOR - NEW APPLICATION

Attorney Docket No.: 1759/2

IN THE UNITED STATES PATENT AND TRADEMARK OFFICE

JC833 U.S. PTO  
09/58619  
06/02/00

In RE Application of: JOSEPH NIR, BARUCH SHAYEVICH AND HANOCH COHEN

Filed Concurrently Herewith

For: COHERENT PROCESSING OF SATELLITE SIGNALS TO LOCATE A MOBILE UNIT

VERIFIED STATEMENT UNDER 37 CFR 1.27  
CLAIMING STATUS AS A SMALL ENTITY

To The Commissioner of Patents and Trademarks:

As a below named Inventor, I hereby declare that:

I qualify as an independent inventor as defined in 37 CFR 1.9(c) for purposes of paying reduced fees under 35 USC § 41(a) and § 41(b) to the Patent and Trademark Office with regard to the above-entitled invention described in the specification filed herewith.

I have not assigned, granted, conveyed or licensed and am under no obligation under contract or law to assign, grant, convey or license, any rights in the invention to any party who could not qualify as a small entity under 37 CFR 1.9(f), namely any person who could not be classified as an independent inventor under 37 CFR 1.9(c) if that person had made the invention, or any concern which would not qualify as a small business concern under 37 CFR 1.9(d) or a nonprofit organization under 37 CFR 1.9(e).

Each party, if any, who could qualify as a small entity under 37 CFR 1.9(f) and to whom I have assigned, granted, conveyed or licensed or am under an obligation under contract or law to assign, grant, convey or license, any rights in the invention is listed below:

Full Name (Party 1) : CELLGUIDE LTD.

Address : BERGMAN 2, 76705 REHOVOT, ISRAEL

Status : ☐ Individual ☒ Small Business Concern ☐ Nonprofit Organization

Full Name (Party 2) : \_\_\_\_\_

Address : \_\_\_\_\_

Status : ☐ Individual ☐ Small Business Concern ☐ Nonprofit Organization

I acknowledge the duty under 37 CFR 1.28(b) to file, in this application, notification of any change in status resulting in loss of entitlement to small entity status prior to paying, or at the time of paying, the issue fee due after the date on which status as a small entity is no longer appropriate.

I hereby declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code, and that such willful false statements may jeopardize the validity of the application and any patent issuing thereon.

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Date: 15 DEC 99

## APPLICATIONS FOR PATENT

Inventors: Joseph Nir, Baruch Shayevich and Hanoch Cohen

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Title: COHERENT PROCESSING OF SATELLITE SIGNALS TO  
LOCATE A MOBILE UNIT

10 FIELD AND BACKGROUND OF THE INVENTION

The present invention relates to positioning systems for portable radio transceivers, and, more particularly, to a positioning system for a mobile unit of a wireless communications network, based on signals from a satellite network such as the Global Positioning System (GPS) and on signals from the base stations of the  
15 wireless communications network.

The Global Positioning System is a system of low Earth orbit satellites that transmit standard signals that can be used to establish the location of a user equipped with a suitable GPS receiver. For civilian applications, the signals are "C/A codes" that are pseudorandom noise (PN) sequences transmitted at a chip rate of 1023 KHz  
20 and a repetition period of 1023 chips, so that each frame of the PN sequence includes 1023 chips. Each satellite is assigned its own PN sequence, and the PN sequences of the various satellites are mutually orthogonal.

Superposed on the PN sequences transmitted by a GPS satellite is a Satellite Data Message, transmitted every 30 seconds at a rate of 50 bits per second. Each bit  
25 (+1 or -1) of the Satellite Data Message is modulated onto the satellite's signal by multiplying 20 consecutive frames of the PN sequence by the bit. The first 900 bits of the Satellite Data Message includes the satellite ephemeris and time model for the respective satellite. The remaining 600 bits of the Satellite Data Message include a portion of the GPS almanac, which is a 15,000 bit block of coarse ephemeris and time

model data for the entire GPS system. In addition, bits 1-8, bits 301-308, bits 601-608, bits 901-908 and bits 1201-1208 of every Satellite Data Message are identical 8-bit (160 millisecond) headers that are invariant in time and that are identical for all the GPS satellites; and bits 31-60, bits 331-360, bits 631-660, bits 931-960 and bits 1231-1260 of every Satellite Data Message are 30-bit (600 millisecond) handover words that are time-variant (these handover words include representations of the time of the week), but that, like the headers, are identical for all the GPS satellites.

Conventionally, a GPS receiver acquires and tracks the signals from at least four GPS satellites, by correlating the received signal with the satellites' respective PN sequences and locking on to the correlation peaks. Once the satellites are acquired and tracked, the GPS receiver decodes the ephemeris and time model, for each acquired satellite, from the respective Satellite Data Messages. These models include sufficient ephemeris data to enable the GPS receiver to compute the satellite's positions. The correlation peaks obtained during the continued tracking of the satellites provide measured times of arrival of these PN sequence frames. The differences between an arbitrary reference time and measured times of arrival, multiplied by the speed of light, are pseudoranges  $\rho$  from the satellites to the GPS receiver. Typically, the reference time is the time at which the satellites commenced transmission of their respective PN sequences, as measured by the GPS receiver clock, which in general is offset from the GPS system clock by an unknown time offset. A pseudorange  $\rho$  is related to the true range  $R$  of the respective satellite by  $\rho=R+c_b$ , where the range offset  $c_b$  is the time offset,  $T_0$ , of the GPS receiver relative to GPS system time, multiplied by the speed of light  $c$ :  $c_b=T_0c$ . From these pseudoranges, and from the known positions of the satellites as functions of time, the position of the GPS

receiver is calculated by triangulation. Pseudoranges to at least four satellites are needed to solve at least four simultaneous equations of the form

$$|\mathbf{s}-\mathbf{r}| = \rho - c_b$$

where  $\mathbf{s}$  is the position vector of a satellite and  $\mathbf{r}=(x,y,z)$  is the position vector of the GPS receiver, for the three unknown Cartesian coordinates  $x, y, z$  of the GPS receiver and for  $c_b$ . The satellites are sufficiently far from the GPS receiver that these equations can be linearized in  $x, y$  and  $z$  with no loss of accuracy.

Several methods are known for increasing the efficiency with which a GPS receiver establishes its position and for reducing the power requirements of a GPS receiver. Schuchman et al., in U. S. Patent No. 5,365,450, which is incorporated by reference for all purposes as if fully set forth herein, teach the provision of the Satellite Data Messages to a GPS receiver integrated into a mobile unit, such as a cellular telephone, of a wireless communications network such as a cellular telephone network, by transmitting the Satellite Data Messages to the mobile unit from a base station of the network via the control channel of the network. The Satellite Data Message is 30 seconds long, so even under ideal reception conditions with parallel processing of the signals from all the satellites in view, it necessarily takes more than 30 seconds to get a GPS position fix. Prior knowledge of the Satellite Data Message reduces this time to under 10 seconds.

Krasner, in U. S. Patent No. 5,663,734, which is incorporated by reference for all purposes as if fully set forth herein, teaches a GPS receiver for a mobile unit in which, as in the GPS receiver of Schuchman et al., the Satellite Data Message is obtained by a wireless link to a base station, but then, instead of processing GPS signals in real time, the GPS receiver stores up to one second's worth of signals (1000

PN sequence frames per satellite), along with the initial time of arrival of the signals, and processes the stored signals. Groups of 5 to 10 frames each are summed and correlated with the PN sequences of satellites expected to be in view, and the resulting correlation functions are added incoherently. The summation over up to 1000 frames  
 5 boosts the signal to noise ratio by a corresponding amount, and the post facto processing requires much less power than real time processing.

Duffet-Smith et al., in PCT Application WO 99/21028, which is incorporated by reference for all purposes as if fully set forth herein, teach a system and method for locating a mobile receiver ("remote unit") of a wireless communications network.

10 The remote unit and a base unit at a fixed location both receive signals on a control channel from three or more base transceiver stations (BTSs), and both correlate invariant portions of the control channel signals. Compact descriptions of the correlation peaks, such as low order polynomial fits, are transmitted to a central processor by both the base unit and the remote unit. The central processor recovers,  
 15 from the correlation peaks, the time offsets of the BTS transmissions to the base and remote units. These time offsets are the functional equivalents of GPS pseudoranges. Given the (fixed) locations of the BTSs and the base unit, the central processor computes the position of the remote unit, essentially by triangulation, and transmits that location to the remote unit.

20 A GPS positioning system is predicated on the GPS receiver having a clear line of sight to at least four GPS satellites. This often is not the case in an urban environment. Urban environments commonly have cellular telephone networks installed, so in principle the system taught by Duffet-Smith et al. could be used for locating a mobile transceiver configured as a cellular telephone; but cellular

telephones generally are in simultaneous contact with at most two base stations. Sheynblatt, in U. S. Patent No. 5,999,124, Camp, in PCT Application WO99/61934, and Watters et al., in U. S. Patent No. 5,982,324, all teach methods of using signals from both GPS satellites and terrestrial BTSs to determine the location of a mobile receiver. These prior art methods generally rely on triangulation, as described above, using signals received simultaneously from a combined total of four or more GPS satellites and BTSs. As noted above, usually, at most two BTSs are "in view" simultaneously; and in a sufficiently cluttered and noisy urban environment, useable signals may be available from only one or two GPS satellites at any one time. These prior art methods also rely on all the BTSs being mutually synchronized, at least with each other if not with the GPS system. Such synchronization is optional under the GSM cellular standard. There is thus a widely recognized need for, and it would be highly advantageous to have, a system and method for locating a mobile unit of a wireless communications network, for example a mobile cellular telephone in a cluttered and noisy urban environment, that derives the location of the mobile unit from signals received sequentially from low Earth orbit satellites such as GPS satellites and from base stations of the communications network, without requiring that the BTSs be mutually synchronized.

## SUMMARY OF THE INVENTION

According to the present invention there is provided a method for determining a pseudorange, and a rate of change thereof, to a beacon that transmits a signal including a plurality of blocks, each block including a plurality of frames of a pseudonoise sequence, each block being multiplied by a bit of a data sequence, the



signal being shifted relative to a nominal frequency by a frequency shift, the method including the steps of: (a) receiving the signal; (b) digitizing the received signal, thereby producing a digitized signal including a plurality of bits; (c) arranging the digitized signal as columns of an input matrix that includes a plurality of rows, each  
 5 column including consecutive the bits of the digitized signal that correspond to an integral number of frames of the pseudonoise sequence; and (d) performing a discrete orthogonal transform on each row of the input matrix, thereby producing a transformed matrix.

According to the present invention there is provided a method for determining  
 10 a pseudorange, and a rate of change thereof, to each of a plurality of beacons, each beacon transmitting a respective signal including a plurality of blocks, each block including a plurality of frames of a pseudonoise sequence, each block being multiplied by a bit of a data sequence, the pseudonoise sequences and the data sequences being uniquely associated with respective satellites, all the pseudonoise sequences being of  
 15 equal length, the pseudonoise sequences being mutually orthogonal, the method including the steps of: (a) receiving the transmitted signals collectively as a received signal; (b) digitizing the received signal, thereby producing a digitized signal including a plurality of bits; and (c) for each beacon, multiplying groups of the bits of the digitized signal, that correspond to the blocks of the signal transmitted by the each  
 20 beacon, by respective bits of the data sequence of the each beacon.

According to the present invention there is provided a receiver for receiving a signal transmitted by a beacon, the signal including a plurality of frames of a pseudonoise sequence, including: (a) an antenna for receiving the transmitted signal; (b) a mechanism for digitizing the received signal to produce a digitized signal

including a plurality of bits; (c) a memory for storing the digitized signal as columns of a matrix that includes a plurality of rows, each column including consecutive the bits of the digitized signal that correspond to an integral number of frames of the pseudonoise sequence; and (d) a processor for performing a discrete orthogonal transform on each of the rows of the matrix.

According to the present invention there is provided a locator system for locating a mobile unit, including: (a) at least one beacon having a respective pseudonoise sequence and a respective data sequence, each data sequence including a plurality of bits, each at least one beacon operative to transmit a respective transmitted signal, each respective transmitted signal including a plurality of blocks, each block including a plurality of frames of the respective pseudonoise sequence, each block being multiplied by a bit of the respective data sequence, (b) a reference unit including: (i) a reference unit receiver for: (A) receiving the at least one transmitted signal as a reference unit received signal, and (B) recovering the at least one data sequence from the respective reference unit received signal, and (ii) a transmitter for transmitting the at least one data sequence to the at least one mobile unit; and (c) in the mobile unit: (i) a first mobile unit receiver for receiving the at least one data sequence, and (ii) a second mobile unit receiver including: (A) an antenna for receiving the at least one transmitted signal collectively as a mobile unit received signal, (B) a mechanism for digitizing the mobile unit received signal to produce a digitized signal including a plurality of bits, (C) a memory for storing, for each of the at least one beacon, an instance of the digitized signal, and (D) a processor for multiplying each of the at least one instance of the digitized signal by the bits of the respective data sequence.

According to the present invention there is provided a method for determining a location of a receiver, including the steps of: (a) providing a plurality of beacons having respective pseudonoise sequences, all the pseudonoise sequences being of equal length; (b) transmitting, by each beacon, a respective signal including a plurality of frames of the respective pseudonoise sequence; (c) receiving the transmitted signals collectively as a received signal, by the receiver; (d) inferring, for each beacon, a pseudorange and a rate of change of the pseudorange; and (e) inferring the location of the receiver from the pseudoranges and from the rates of change of the pseudoranges.

According to the present invention there is provided a method for determining a pseudorange, and a rate of change thereof, to a beacon that transmits a signal including a plurality of blocks, each block including a plurality of frames of a pseudonoise sequence, each block being multiplied by a bit of a data sequence, the signal being shifted relative to a nominal frequency by a Doppler frequency shift, the method including the steps of: (a) receiving the signal; (b) digitizing the received signal, thereby producing a digitized signal including a plurality of bits; and (c) applying a matched filter algorithm to the digitized signal to extract therefrom the pseudorange and the rate of change of the pseudorange, the matched filter algorithm including: (i) demodulating the digitized signal relative to the data sequence.

According to the present invention there is provided a method for locating a mobile unit, including the steps of: (a) providing a plurality of asynchronous stationary beacons, each stationary beacon transmitting a respective stationary beacon signal; (b) providing a plurality of mobile beacons moving on known respective trajectories, each mobile beacon transmitting a respective mobile beacon signal, all the mobile beacons being mutually synchronized; (c) for each of at least one of the

stationary beacons: (i) receiving the respective stationary beacon signal, by the mobile unit, and (ii) inferring, from the respective stationary beacon signal, at least one parameter related to a range to the each stationary beacon; (d) for each of at least one of the mobile beacons: (i) receiving the respective mobile beacon signal, by the mobile unit, and (ii) inferring, from the respective mobile beacon signal, at least one parameter related to a range to the each mobile beacon; (e) synchronizing the mobile unit and the stationary beacons to the mobile beacons by steps including: (i) providing, for each stationary beacon, a reference unit for measuring time offsets between the mobile beacons and the each stationary beacon, (ii) measuring the time offsets, by the reference units, and (iii) transmitting at least one the time offset from a respective the reference unit to the mobile unit, the inferring of the parameters being based on the synchronization; and (f) inferring, from the parameters, a location of the mobile unit.

According to the present invention there is provided a method for tracking a mobile unit, including the steps of: (a) providing a plurality of beacons, each beacon transmitting a respective ranging signal; (b) obtaining an initial estimate of a location of the mobile unit; (c) receiving, by the mobile unit, one of the ranging signals; (d) inferring, from the ranging signal, at least one parameter related to a range to the beacon that transmitted the ranging signal; and (e) obtaining an updated estimate of the location of the mobile unit, based on an initial state estimate that includes the initial estimate of the location and on an update measurement that includes the at least one parameter.

According to the present invention there is provided a locator system for locating a mobile unit, including: (a) a plurality of asynchronous stationary beacons,

each stationary beacon being operative to transmit a respective stationary beacon signal; (b) a plurality of mobile beacons, each mobile beacon operative to transmit a respective mobile beacon signal while moving on a known respective trajectory, all the mobile beacons being mutually synchronized; and (c) in the mobile unit: (i) a

5 mobile unit clock for producing a mobile unit timing signal; (ii) a first mobile unit receiver for receiving and digitizing the stationary beacon signals; (iii) a second mobile unit receiver for receiving and digitizing the mobile beacon signals, thereby producing corresponding digitized mobile beacon signal; (iv) a memory for storing the digitized mobile beacon signal; (v) a mobile unit processor for transforming each of

10 the stationary beacon signals into a corresponding stationary beacon pseudorange, and for transforming each of the digitized mobile beacon signals into a corresponding mobile beacon pseudorange subsequent to the storing of the each digitized mobile beacon signal in the memory; and (d) at least one reference unit including: (i) a first reference receiver for receiving the mobile beacon signals and inferring therefrom a

15 common start time of the mobile beacon signals, (ii) a second reference unit receiver for receiving at least one of the stationary beacon signals from at least one respective the stationary beacon, (iii) a reference unit clock for producing a reference unit timing signal, (iv) a reference unit processor for inferring, for each at least one respective stationary beacon, a timing offset between a start time of the at least one respective

20 stationary beacon signal and the common start time of the mobile beacon signals; and (iii) a transmitter for transmitting the at least one timing offset; the first mobile unit receiver being operative to receive the at least one timing offset from each at least one reference unit transmitter; the pseudoranges being with reference to both the mobile unit timing signal and the at least one timing offset.

According to the present invention there is provided a locator system for locating a mobile unit, including: (a) a plurality of beacons, each beacon transmitting a respective ranging signal; and (b) in the mobile unit: (i) a mechanism for obtaining an initial estimate of a location of the mobile unit, (ii) a receiver for receiving the ranging signals, (iii) a processor for inferring, from one of the ranging signal, at least one parameter related to a range to the beacon that transmitted the ranging signal, and (iv) an updating mechanism for updating the location estimate, based on an update measurement that includes the at least one parameter.

According to the present invention there is provided a method for tracking a mobile unit, including the steps of: (a) providing a plurality of mobile beacons, each mobile beacon transmitting a respective ranging signal; (b) providing a plurality of stationary beacons, each stationary beacon transmitting a respective ranging signal; (c) obtaining an initial estimate of a location of the mobile unit; (d) receiving, by the mobile unit, a first ranging signal from a respective beacon of one of the pluralities of beacons; (e) inferring, from the first ranging signal, at least one parameter related to a range to the beacon that transmitted the first ranging signal; (f) obtaining a first updated estimate of the location of the mobile unit, based on the initial location and on the at least one parameter that is related to the range to the beacon that transmitted the first ranging signal; (g) receiving, by the mobile unit, a second ranging signal from a respective beacon of another of the pluralities of beacons; (h) inferring, from the second ranging signal, at least one parameter related to a range to the beacon that transmitted the second ranging signal; and (i) obtaining a second updated estimate of the location of the mobile unit, based on the first updated location and on the at least

one parameter that is related to the range to the beacon that transmitted the second ranging signal.

According to the present invention there is provided a locator system for locating a mobile unit, including: (a) a plurality of mobile beacons, each mobile  
5 beacon transmitting a respective ranging signal; (b) a plurality of stationary beacons, each stationary beacon transmitting a respective ranging signal; and (c) in the mobile unit: (i) a mechanism for obtaining an initial estimate of a location of the mobile unit, (ii) a receiver for receiving the ranging signals, (iii) a processor for inferring, from each of the ranging signals, at least one parameter related to a range to the beacon that  
10 transmitted the ranging signal, and (iv) an updating mechanism for updating the initial location estimate, using the at least one parameter that is related to the range to a beacon of one of the pluralities of beacons, thereby providing a first updated estimate of the location of the mobile unit, and for updating the first updated estimate, using the at least one parameter that is related to the range to a beacon of another of the  
15 pluralities of beacons.

In the broadest sense, the present invention includes methods and systems that enable a mobile unit to determine its location by appropriate processing of signals received from mobile beacons and stationary beacons. As understood herein, a  
20 beacon is a device that includes a transmitter that transmits a signal, typically a radio frequency signal, that the mobile unit receives and then uses to determine its own location. Some of the beacons are stationary, meaning that these beacons are in fixed and known locations. Other beacons are mobile, meaning that these beacons move along known trajectories, such that at any time, the instantaneous locations of the relevant mobile beacons are known. Typically, the mobile unit is a cellular telephone,

the stationary beacons are base transceiver stations of a cellular telephone network, and the mobile beacons are satellites of a low-Earth-orbit-satellite-based positioning system such as the GPS system and the Russian GLONASS system, the trajectories of the mobile beacons then being the orbits of the satellites, as described by the satellite ephemeris and time models. More generally, however, the scope of the present invention includes any suitable beacons. For example, the stationary beacons may be special purpose beacons set up specifically to enable the tracking of a mobile unit, and the mobile beacons may be aircraft flying along known trajectories.

Although the scope of the present invention extends to the use of any suitable types of mobile and stationary beacons, the present invention is described herein in terms of the preferred and most typical mobile and stationary beacons: GPS satellites and base stations (BTSs) of a cellular telephone network. It will be clear to those skilled in the art how to adapt the principles of the present invention to other types of mobile and stationary beacons.

The mobile unit receives timing signals from the mobile and stationary beacons. In the case of the mobile beacons being GPS satellites, the timing signals are successive PN frames, modulated by the Satellite Data Messages. In the case of the stationary beacons being BTSs, the timing signals are synchronization burst sequences transmitted in the cellular network control channel.

In order for these signals to be interpreted correctly by the mobile unit, if the cellular telephone network is asynchronous, with the BTSs being synchronized neither to the GPS network nor to each other, then the time offsets between the individual BTSs and the GPS system must be known by the mobile unit. Therefore, at least one reference unit is provided, for determining offsets between the BTSs and GPS time.



One reference unit is associated with one or more BTSs. Each reference unit includes a GPS-based time reference module for receiving the PN frames from the GPS satellites and cross-correlating them with local copies of the PN sequences to determine GPS time, a cellular communications module for receiving the synchronization burst sequences from the associated BTSs, and a synchronization module for determining the timing offsets of the associated BTSs relative to GPS time. The cellular telephone network is provided with a locator service center, and the reference units transmit the timing offsets to the mobile unit via the locator service center. Optionally, one of the reference units is designated as a master reference unit that also transmits the Satellite Data Messages to the mobile unit via the locator service center. (This master reference unit may be physically integrated with the locator service center.) The mobile unit is synchronized to the cellular network in the conventional manner. This allows the mobile unit to also be synchronized approximately to the GPS network: having received the BTS timing offsets from the reference units via the locator service center, the mobile unit computes the time offset between its own clock and GPS time to within about 30 microseconds.

The mobile unit receives the satellite timing signals collectively from the GPS satellites that are in view, digitizes the combined signal, and applies matched filter algorithms to the combined signal to extract pseudoranges and rates of change of pseudoranges from the combined signal, as follows. Successive frames of the digitized combined signal are arranged as columns of a matrix. As many instances of the matrix are formed as there are satellites in view. That the mobile unit is approximately synchronized with the GPS network allows each matrix instance to be demodulated, with respect to the Satellite Data Message of the corresponding satellite,

by multiplying groups of columns of the matrix instance by corresponding bits of the Satellite Data Message of the corresponding satellites. If a master reference unit is provided, then any PN frames may be used to form the instances of the matrix; otherwise, only those PN frames, that correspond to bits of the Satellite Data

5 Messages that are common to all satellites, are used. The rows of the matrices are transformed using a discrete orthogonal transform, preferably a discrete Fourier transform. The elements of the matrices are multiplied by respective Doppler compensation factors. The columns of the matrices are convolved with the PN sequences of the in-view satellites to provide, for each in-view satellite, a signal

10 ambiguity function matrix. The squared magnitudes of the elements of these matrices include local peaks, the row and column coordinates of which correspond to candidate hypotheses for the pseudoranges to the corresponding satellites and to the rates of change of these pseudoranges. This procedure is repeated several times to provide several signal ambiguity function matrices for each in-view satellite. Then, for each

15 in-view satellite, the best hypothesis is chosen, for example, by integrating the matrices noncoherently and then identifying the most likely peak in the sum; or by passing the coordinates of all the peaks of the matrices to a "track before detect" algorithm. In either case, the peaks are identified using a constant false alarm rate detection scheme.

20 The pseudoranges to stationary beacons are simply the differences between the measured times of arrival of the synchronization burst sequences and the transmission times of the synchronization burst sequences according to the GPS system clock, multiplied by the speed of light. Note that, because the mobile unit has received the time offsets between the BTS clocks and the GPS system clock, from the reference

units via the locator service center, the mobile unit can refer these transmission times to the GPS system clock. Note also that, as a consequence, the pseudoranges to the BTSs are with reference to the same range offset,  $c_b$ , as applies to the GPS satellites. As each stationary beacon pseudorange is acquired by the mobile unit, and as each

5 mobile beacon pseudorange - pseudorange rate pair is acquired by the mobile unit, these data are input as update measurements to a dynamic estimator to produce an updated estimate of the location of the mobile unit. Preferably, the dynamic estimator is based on a linear optimal estimator, such as a Kalman filter. The dynamic estimator requires an initial estimate of the location of the mobile unit. This initial estimate is

10 obtained by triangulation. Alternatively, under certain conditions, the location of a predefined reference point, such as the nearest stationary beacon, may be used as the initial estimate of the mobile unit's location.

#### BRIEF DESCRIPTION OF THE DRAWINGS

15 The invention is herein described, by way of example only, with reference to the accompanying drawings, wherein:

FIG. 1 is a block diagram of a mobile unit of the present invention;

FIG. 2 is a schematic diagram of a locator system of the present invention;

FIG. 3 is a block diagram of a reference unit of the present invention;

20 FIG. 4 is a flow chart of the joint estimation of position from satellite pseudoranges, rates of change of satellite pseudoranges and base station pseudoranges.

## DESCRIPTION OF THE PREFERRED EMBODIMENTS

The present invention is of a system and method for locating a mobile unit, for example a mobile unit of a wireless communications system, based on timing signals, both from mobile beacons such as GPS satellites and from stationary beacons such as base stations of the wireless communications system. Specifically, the present invention can be used to locate a suitably equipped cellular telephone in an environment, such as an urban environment, in which it is difficult to establish simultaneous lines of sight to four or more GPS satellites.

Although the present invention is described herein in reference to the GPS system and a cellular telephony network, it will be appreciated that this description is purely illustrative, and that the scope of the present invention extends to the use of any suitable types of mobile and stationary beacons, as defined above.

The principles and operation of a locator system and method according to the present invention may be better understood with reference to the drawings and the accompanying description.

Referring now to the drawings, Figure 1 is a partial high level block diagram of a mobile unit **10** of the present invention. Mobile unit **10** includes an antenna **12** for receiving RF signals transmitted by GPS satellites, a downconverter **14** for downconverting the RF signals received from the GPS satellites to an intermediate frequency (IF), a digitizer **16** for digitizing the IF signals, a clock **22** for providing local timing signals, a memory **18** for storing the digitized GPS signals, and a GPS processor **20** for processing the signals stored in memory **18**. Mobile unit **10** also includes an antenna **24** for exchanging RF signals with BTSs of a cellular telephone network, a cellular modem **26** for downconverting and digitizing the control channel

RF signals received from the BTSs, a memory 27 for storing the digitized BTS signals, and a cellular processor 28 for processing the digitized BTS signals with reference to timing signals from clock 22, as described below. The outputs of GPS processor 20 and cellular processor 28 are reconciled in a joint estimator 30, as described below, to produce a joint estimate of the location of the mobile unit of which mobile unit 10 is a part. GPS processor 20, cellular processor 28 and joint estimator 30 function under the overall supervision of a management unit 29. In particular, management unit 29 formulates requests for supplemental data, and, using cellular modem 26 and antenna 24, transmits these requests as RF signals to the BTSs, as described below.

Figure 2 is a schematic diagram of a locator system 40 of the present invention. Locator system 40 is based on a conventional cellular telephone network 42 that includes multiple BTSs 44, and on the GPS system, represented in Figure 2 by satellites 70. The terrestrial component of locator system 40 adds to cellular telephone network 42 a locator service center 46 and a multitude of reference units 50, at fixed and known locations. Each reference unit 50 is associated with one or more BTSs 44, and communicates with those BTSs 44 and, via the nearest of those BTSs 44, with service center 46. Mobile unit 10, using antenna 24 and cellular modem 26, also receives messages from service center 46 via the nearest BTS 44, as described below.

Figure 3 is a high level block diagram of reference unit 50. Reference unit 50 includes a cellular communications module (CCM) 52 for communicating with service center 46 via cellular network 42, a GPS-based time reference module 66, and a synchronization module 60 for synchronizing CCM 52 to GPS-based time reference

module **66** as described below. CCM **52** includes both transmission functionality **54** and reception functionality **56**, via an antenna **58**. GPS-based time reference module **66** receives GPS satellite signals via an antenna **68**. Synchronization module **60** includes both a clock **64** for providing reference timing signals and a processor **62** for

5 establishing timing offsets between the GPS system and the BTSs **44** that are associated with reference unit **50**, as described below. The main function of each reference unit **50** is to determine these timing offsets and to transmit these timing offset to locator service center **46**. Optionally, one of reference units **50** is designated as a master reference unit. The master reference unit has the additional function of

10 receiving the Satellite Data Messages from satellites **70** and transmitting these Satellite Data Messages to locator service center **46**.

Locator service center **46** is a node of cellular telephone network **42** whose function is to coordinate communications between mobile unit **10** and cellular telephone network **42** that relate to the self-location functionality of mobile unit **10**.

15 In particular, locator service center **46**, upon receipt of a query from mobile unit **10**, transmits to mobile unit **10** supplemental data such as the timing offsets between the GPS system and one or more BTSs **44** and optionally the Satellite Data Messages of in-view satellites **70**. Optionally, the functionality of joint estimator **30** may reside in location service center **46**, with the outputs of GPS processor **20** and cellular

20 processor **28** being transmitted to location service center **46** for reconciliation.

The signal received by mobile unit **10** from the GPS system, as a function of time  $t$ , is a sum of the signals  $G(t)$  transmitted by GPS satellites **70** that are in view at time  $t$ . The signal received by mobile unit **10** from any one satellite **70**, indexed by an index  $j$ , is in the form

$$G_j(t) = K_j D_j(t - \tau_j) g_j(t - \tau_j) \exp[i\omega(t - \tau_j)]$$

where  $i$  is the square root of  $-1$ ;  $\tau_j$  is the true one-way propagation time of radio waves from the satellite to mobile unit **10**;  $K_j$  is an amplitude factor that depends on the true range  $R_j = c\tau_j$  to satellite  $j$  at time  $t$ , and on the GPS antenna gain pattern;  $D_j(t - \tau_j)$  is the Satellite Data Message of satellite  $j$  at time  $t - \tau_j$ ;  $g_j(t - \tau_j)$  is the PN sequence of satellite  $j$  at time  $t - \tau_j$ ; and  $\exp[i\omega(t - \tau_j)]$  is a Doppler factor whose Doppler frequency  $\omega$  depends on the radial component of the velocity of satellite  $j$  relative to mobile unit **10**, on the bias of clock **22**, and on other imperfections in mobile unit **10**. The total signal  $G(t)$  received by mobile unit **10** is the sum of these signals,  $\sum_j G_j(t)$ .

The object of the processing in GPS processor **20** is to measure  $\rho_j = R_j + c_b$  and its rate of change  $d\rho_j/dt$  for all the satellites in view. Note that all the pseudoranges  $\rho_j$  are referred to the same range offset  $c_b$ . As in the prior art methods,  $c_b$  is one of the unknowns that is solved for subsequently to determine the location of mobile unit **10**, so it is not necessary to measure  $\tau_j$  explicitly.

As in US 5,663,734, several hundred seconds worth of a digitized total GPS signal is stored in memory **18** and then processed by GPS processor **20**. One difference between US 5,663,734 and the GPS processing of the present invention is that according to the present invention, groups of many more than 5 to 10 PN sequence frames are processed coherently.

The total signal received by mobile unit **10** is oversampled by digitizer **16**, starting at an initial reception time  $t_0$ , typically at a sampling rate of between two and four samples per chip. With four samples per chip, there are a total of  $4 \times 1023 = 4092$  samples per frame. For each in-view satellite **70**, indexed by index  $j$ , these samples

are stored as successive columns of a matrix  $\mathbf{M}_{ij}$  that, in the case of a sampling rate of four samples per chip, has 4092 rows and as many columns as the number of consecutive frames included in the computation: the first column of  $\mathbf{M}_{ij}$ , matrix elements  $M_{ij}(1,1)$  through  $M_{ij}(4092,1)$  are the first 4092 samples, corresponding to the first millisecond's worth of data; the second column of  $\mathbf{M}_{ij}$ , matrix elements  $M_{ij}(1,2)$  through  $M_{ij}(4092,2)$ , are the second 4092 samples, corresponding to the second millisecond's worth of data; etc., through a number of columns  $n$  equal to the number of frames of data that are stored in memory **18**, at one millisecond per frame. Preferably, between 100 milliseconds and 1000 milliseconds worth of data (100 to 1000 frames) are stored in memory **18** as matrix  $\mathbf{M}_{ij}$  for each in-view satellite **70**.

The first processing step is to demodulate each matrix  $\mathbf{M}_{ij}$ , *i.e.*, to remove the influence of Satellite Data Message  $D_j$ . This is done by multiplying groups of columns of  $\mathbf{M}_{ij}$  by the corresponding bits of  $D_j(t-\tau_j)$ . A necessary prerequisite for this demodulation is for the bits of  $D_j$  to be aligned with the columns of  $\mathbf{M}_{ij}$  to within about one millisecond. This alignment is referred to herein as "approximate synchronization" of mobile unit **10** with the GPS system.

Cellular telephone network **42** may be synchronized to the GPS system clock in any case, for example, if cellular telephone network operates according to the CDMA IS-95 standard. Mobile unit **10** is synchronized to cellular telephone network **42** in the conventional manner. Because cellular telephone network **42** is time-aligned (synchronized) with GPS system time, mobile unit **10** also is synchronized to GPS system time and therefore is able to demodulate matrix  $\mathbf{M}_{ij}$  relative to Satellite Data Message  $D_j$ .



If cellular telephone network **42** is not synchronized to the GPS system clock, and, in particular, if BTSs **44** are not synchronized to each other, mobile unit **10** nevertheless can obtain, from reference units **50** via locator service center **46**, the transmission times of the synchronization burst sequences of BTSs **44** according to the GPS clock. When CCM **52** of a reference unit **50** receives a synchronization burst sequence from one of the BTSs **44** associated with reference unit **50**, synchronization module **60** determines the time offset between that BTS **44** and the GPS system with reference to clock **64**. Because the locations of reference unit **50** and the associated BTSs **44** are fixed and known, the propagation time from each associated BTS **44** to reference unit **50** also is known, so the time offset is referred to the time at which that BTS **44** transmitted the synchronization burst sequence, and not the time at which reference unit **50** received the synchronization burst sequence. CCM **52** then transmits this time offset to locator service center **46**. This is done for the synchronization burst sequence of each associated BTS **44**, so that locator service center **46** always has a current list of time offsets of all BTSs **44** from the GPS system. Finally, mobile unit **10** receives the current time offsets of the relevant BTSs **44** from locator service center **46** on request. The time offset of the nearest BTS **44** from the GPS system is used by mobile unit **10** to correct the nominal transmission times of the synchronization burst sequences from the nearest BTS **44** for this offset, and so to obtain an estimate of its own time offset  $T_0$  from GPS system time to within 30 microseconds. The time offsets of BTSs **44** from the GPS system also are used to obtain BTS pseudoranges that are referred to range offset  $c_b$ , as described below.

If the frames that are used to construct matrix  $M_{ij}$  correspond to Satellite Data Message header bits or to Satellite Data Message handover word bits, which are

common to the Satellite Data Messages of all satellites **70**, then mobile unit **10** either knows these bits *a priori* (in the case of header bits) or computes these bits based on the time of the week (in the case of handover word bits). In the embodiment of the present invention in which a master reference unit is designated to receive the Satellite

5 Data Messages and relay these Satellite Data Messages to locator service center **46**, mobile unit **10** sends a request to locator service center **46** to obtain Satellite Data Message  $D_j$  so that the satellite-dependent bits of Satellite Data Message  $D_j$  can be used to demodulate matrix  $M_{fj}$ .

The second processing step is to transform each matrix  $M_{fj}$  into a matrix  $M_{fij}$

10 by performing a discrete orthogonal transform on each of the rows of matrix  $M_{fj}$ , thereby transforming the rows from the time domain to the frequency domain, so that matrices  $M_{fij}$  are in a mixed time-frequency domain. The most preferred discrete orthogonal transform is the discrete Fourier transform, although the scope of the present invention includes other transforms, such as the Walsh-Hadamard transform,

15 the various kinds of Wavelet transforms and the Hartley transform.

The third processing step is the multiplication of the elements of each matrix  $M_{fij}$  by Doppler compensation factors. The form of the Doppler compensation factor for the  $(r,d)$  element of matrix  $M_{fij}$  is  $\exp\left(-\frac{2\pi}{m}ir\left(\frac{d}{n}+h\right)\right)$ , where  $m$  is the number of rows in matrix  $M_{fij}$ ,  $n$  is the number of columns in matrix  $M_{fij}$  and  $h$  is the integer part

20 of the Doppler frequency shift of the signal  $G_j(t)$ , measured in KHz. Note that, with one one-millisecond PN frame per column of matrix  $M_{fj}$ , each row of matrix  $M_{fij}$  spans one KHz. Initially, the true Doppler frequency shift of each satellite **70** can be estimated by mobile unit **10** to within about 4 KHz. Therefore, four different sets of

Doppler compensation factors are needed, resulting in four different matrices  $\mathbf{M}_{ij}$  for each satellite  $j$ . Subsequently, as described below, much more accurate estimates of the Doppler frequency shifts of satellites 70 are obtained, and only one matrix  $\mathbf{M}_{ij}$  is needed per satellite  $j$ . For clarity, the remaining discussion of how GPS pseudoranges and pseudorange rates of change are measured according to the present invention is in terms of a single matrix  $\mathbf{M}_{ij}$  per satellite  $j$ .

The fourth processing step is to convolve each of the columns of each matrix  $\mathbf{M}_{ij}$  with the PN sequence of satellite  $j$ , to produce a matrix  $\mathbf{M}_j$  that is equivalent to the matched filter, in range and frequency, known in the field of radar signal processing as the “signal ambiguity function”. See, for example, Fred E. Nathanson, *Radar Design Principles, Second Edition* (McGraw-Hill, 1991). It is this use of demodulated signal ambiguity functions, subsequent to approximate synchronization of mobile unit 10 to the GPS system, that enables the present invention to process coherently many more than the 5 to 10 PN sequence frames that are processed coherently by Krasner in US 5,663,734. The preferred convolution method is the well-known fast convolution method, also used by Krasner in US 5,663,734: perform a discrete Fourier transform on a column of matrix  $\mathbf{M}_{ij}$ , multiply the transformed column by the discrete Fourier transform of the PN sequence, and invert the discrete Fourier transform of the column. Note that the multiplication of the elements of matrix  $\mathbf{M}_{ij}$  by the satellite-dependent parts of the Doppler compensation factors,  $\exp\left(-\frac{2\pi irh}{m}\right)$ , may be deferred to this processing step and implemented by, for each transformed column of  $\mathbf{M}_{ij}$ , appropriately shifting the discrete Fourier transform of the PN sequence prior to the multiplication of the transformed column by the discrete

Fourier transform of the PN sequence. Note also that the matrix elements of  $M_j$  are complex. The overall appearance of matrix  $M_j$ , in terms of the squared magnitudes of its elements, is a set of peaks above a general background level. The row coordinate of a peak is its pseudorange (units of distance) and the column coordinate (units of frequency) of a peak is proportional to the rate of change of its pseudorange (units of velocity).

Although the preferred structure of matrix  $M_{ij}$  has one frame per column, it also is possible to construct matrix  $M_{ij}$  with more than one frame per column, as long as an integral number of frames are used per column. Using one frame per column gives the optimal resolution for determining the rates of change of the pseudoranges.

A constant false alarm rate detection scheme is used to identify the peaks. Preferably, the detection scheme is based on an adaptive threshold, based on a background level that is obtained, matrix element by matrix element, by averaging the squared magnitudes of all the matrix elements in a window (for example, an 8x8 window or a 16x16 window) around the target matrix element. The set of matrix elements, whose squared magnitudes exceed their local average by a threshold factor, is in the form of isolated clusters of matrix elements, with each cluster corresponding to one peak. The coordinates of the peaks are the centroids of the clusters, weighted by the magnitudes of the matrix elements in the clusters. The threshold factor is determined by the required maximum false alarm rate, as is well-known in the art.

The output of this procedure is a set of hypotheses for the pseudorange  $p_j$  and its rate of change  $dp_j/dt$  associated with the initial reception time  $t_0$ . At this point, the hypothesis associated with the peak most likely to be the most reliable measure of  $p_j$  and  $dp_j/dt$ , for example, the highest peak, could be selected to provide the coordinates

that determine  $\rho_j$  and  $d\rho_j/dt$ . Preferably, however, several instances of the matrices  $\mathbf{M}_j$ , based on several successive instances of the digitized total GPS signal, are generated. Preferably, between five and ten such instances of each matrix  $\mathbf{M}_j$  are so generated. Subsequent processing is directed towards obtaining, for satellite  $j$ , values  
 5 of  $\rho_j$  and  $d\rho_j/dt$  that are based on all the instances of  $\mathbf{M}_j$  processed collectively.

One method of processing the instances of  $\mathbf{M}_j$  collectively is to integrate them noncoherently. In standard noncoherent integration, the matrix elements of all the instances are replaced by their squared magnitudes; the instances are added to each other, matrix element by matrix element, to produce a sum matrix; the false alarm  
 10 detection scheme described above is used to identify the peaks of the sum matrix; and the coordinates of the peak most likely to be the most reliable measure of  $\rho_j$  and  $d\rho_j/dt$ , for example, the highest peak, are taken as the values of  $\rho_j$  and  $d\rho_j/dt$ . In binary noncoherent integration, the matrix elements of all the instances are replaced by their squared magnitudes, but then the constant false alarm detection scheme is  
 15 used to compare each matrix element with a threshold. Matrix elements that exceed the threshold are replaced by ones, and all other matrix elements are replaced by zeros. The instances now are added to each other, matrix element by matrix element, to produce a sum matrix, and a second threshold is applied to the sum matrix to identify its peaks. Again, the coordinates of the peak, of the sum matrix, that is most  
 20 likely to represent the most reliable measure of  $\rho_j$  and  $d\rho_j/dt$ , are used as the values of  $\rho_j$  and  $d\rho_j/dt$ .

Another way to obtain values of  $\rho_j$  and  $d\rho_j/dt$  that are based on all the instances of  $\mathbf{M}_j$  processed collectively is to replace the matrix elements of all the instances with their squared magnitudes, as in noncoherent integration, but then to

apply the constant false alarm rate detection scheme separately to each instance, to obtain, *for each instance*, a set of hypotheses for  $p_j$  and  $dp_j/dt$ . These hypotheses and their associated start times are input to a “track before detect” identification scheme to identify the most likely hypothesis. “Track before detect” is a method that is well known in radar technology for identifying targets in a low signal-to-noise environment. See, for example, the book by Fred E. Nathanson cited above.

The pseudoranges to BTSs 44 are the differences between the measured times of arrival of the synchronization burst sequences and the nominal transmission times of the synchronization burst sequences according to the GPS system clock. If cellular telephone network 42 is synchronized to the GPS system clock, then the nominal transmission times of the synchronization burst sequences, according to clock 22, are offset from their true times by the same time offset,  $T_0$ , as the time offset by which the nominal transmission times of the PN sequences by satellites 70, according to clock 22, are offset from *their* true times. As a result, all the pseudoranges, both to satellites 70 and to BTSs 44, share the same range offset  $c_b$ . If cellular telephone network 42 is not synchronized to the GPS system clock, then mobile unit 10 obtains the time offsets of BTSs 44 from the GPS system from locator service center 46, as described above in the context of the approximate synchronization of mobile unit 10 with the GPS system for the purpose of demodulation. Mobile unit 10 then uses these time offsets to correct the times of arrival of the synchronization burst sequences from BTSs 44, as measured by clock 22, so that the pseudoranges to BTSs 44 are referred to the same range offset  $c_b$  as the pseudoranges to satellites 70. The pseudoranges to the in-view satellites 70, the rates of change of the pseudoranges to the in-view satellites 70, and the pseudoranges to BTSs 44 are input to joint estimator 30.

The method used by joint estimator **30** to combine pseudoranges  $\rho$  and pseudorange rates of change  $dp/dt$  from GPS processor **20**, and estimated pseudoranges  $\rho$  from cellular processor **28** to derive the location of mobile unit **10** now will be discussed. The essence of this method is to use each new measurement (a

5 pseudorange  $\rho$  and pseudorange rate of change  $dp/dt$  relative to a particular satellite **70** or a pseudorange  $\rho$  relative to a particular BTS **44**), as that measurement is obtained, to update an *a priori* estimate of the location vector  $\mathbf{r}=(x,y,z)^T$  and of the velocity vector  $\mathbf{v}=(v_x,v_y,v_z)^T$  of mobile unit **10**. (It is for this reason that this measurement is referred to herein as an "update measurement".) If mobile unit **10**

10 were to remain stationary between successive measurements and if range offset  $c_b$  were to remain constant between successive measurements, or if four or more measurements were to be obtained from matrices  $\mathbf{M}_j$  computed from signals received simultaneously by mobile unit **10** from the GPS system, it would be possible to fix the location of mobile unit **10** by triangulation. Because mobile unit **10** moves and  $c_b$  may

15 drift between successive measurements, if measurements obtained from GPS signals received by mobile unit **10** at different times are to be used, or if pseudoranges to BTSs **44** are to be used, a method of estimating  $\mathbf{r}$  that accounts for the motion of mobile unit **10** and for the drift of clock **22** must be used. Such a method is referred to herein as a "dynamic" estimation method. The preferred dynamic estimation methods

20 belong to the class of linear optimal estimators. The most preferred dynamic estimation method is based on a Kalman filter. This method now will be described in detail.

In-between updates of  $\mathbf{r}$  and  $\mathbf{v}$ , mobile unit **10** is assumed to move in a known manner. The model of the motion of mobile unit **10** used herein as an example is

linear motion with additive random noise. Other models, such as motion constrained to be along a known road, also may be used. According to the linear model of the motion of mobile unit **10**, if the location vector of mobile unit **10** at time  $t_{k-1}$  is  $\mathbf{r}_{k-1}$  and the velocity vector of mobile unit **10** at time  $t_{k-1}$  is  $\mathbf{v}_{k-1}$ , then the location and velocity vectors at time  $t_k = t_{k-1} + \delta_k$  are  $\mathbf{r}_k = \mathbf{r}_{k-1} + \mathbf{v}_{k-1} \delta_k$  and  $\mathbf{v}_k = \mathbf{v}_{k-1}$ . The range bias  $c_b$  is defined as the difference between the time measured by clock **22** and the GPS clocks, multiplied by the speed of light  $c$ , and the range drift  $c_d$  is defined as  $d(c_b)/dt$ . Then  $c_{b(k)} = c_{b(k-1)} + c_{d(k-1)} \delta_k$  and  $c_{d(k)} = c_{d(k-1)}$ . The (eight component) navigation vector  $\mathbf{n}$  is defined as the concatenation of  $\mathbf{r}$ ,  $\mathbf{v}$ ,  $c_b$  and  $c_d$ :  $\mathbf{n} = (x, y, z, v_x, v_y, v_z, c_b, c_d)^T$ .

The state vector  $\mathbf{e} = (\delta x, \delta y, \delta z, \delta v_x, \delta v_y, \delta v_z, \delta c_b, \delta c_d)^T$  of the Kalman filter is the estimated navigation error, i.e., the difference between the true navigation vector and the linearly propagated navigation vector. This state vector obeys the propagation equation

$$\mathbf{e}_k = \mathbf{A}_{k-1} \mathbf{e}_{k-1} + \mathbf{w}_{k-1}$$

where the transition matrix  $\mathbf{A}_{k-1}$  is:

$$\mathbf{A}_{k-1} = \begin{pmatrix} 1 & 0 & 0 & \delta_k & 0 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 & \delta_k & 0 & 0 & 0 \\ 0 & 0 & 1 & 0 & 0 & \delta_k & 0 & 0 \\ 0 & 0 & 0 & 1 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 1 & \delta_k c \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 1 \end{pmatrix}$$

and  $\mathbf{w}_{k-1}$  is the process noise.  $\mathbf{w}_k$  is defined via the model noise covariance matrix

$\mathbf{Q}_{k-1}$ :



$$Q_{k-1} = W_{k-1}W_{k-1}^T = \begin{pmatrix} RW_x & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & RW_y & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & W \end{pmatrix}$$

where  $RW_x$  and  $RW_y$  are the covariances of the process noise in the  $x$  and  $y$  directions (note that mobile unit **10** is assumed to move substantially in the  $(x,y)$  plane), modeled as random walks, and  $W$  is a range drift error covariance, modeled as white noise.

- 5 Immediately following an update of the location vector  $\mathbf{r}$  and of the velocity vector  $\mathbf{v}$  of mobile unit **10**, the state vector  $\mathbf{e}$  is zero by definition. Subsequently, until the next measurement, as the navigation vector  $\mathbf{n}$  is propagated forward in time, the covariance of the state vector, an  $8 \times 8$  matrix  $\mathbf{P}$ , also is propagated forward in time:

$$\mathbf{P}_k = \mathbf{A}\mathbf{P}_{k-1}^+ \mathbf{A}^T + \mathbf{Q}\delta_k$$

- 10 Here, the superscript “+” refers to the *a posteriori* estimate of  $\mathbf{P}$  after the most recent update, and the superscript “-” refers to the *a priori* estimate of  $\mathbf{P}$ , before the next update. The initialization of  $\mathbf{P}$  is discussed below.

- A measurement obtained from a signal from a satellite **70** is a two component vector,  $\mathbf{z}=(\rho, d\rho/dt)^T$ . The matrix  $\mathbf{H}_k$  that relates the state vector  $\mathbf{e}$  to this measurement
- 15 vector at time  $t_k$  is:

$$\mathbf{H}_k = \begin{pmatrix} u_x & u_y & u_z & 0 & 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & u_x & u_y & u_z & 0 & 1 \end{pmatrix}$$

where the vector  $\mathbf{u}=(u_x, u_y, u_z)$  is a unit vector from the estimated location of mobile unit **10** to satellite **70**:

$$\mathbf{u} = (\mathbf{r}_k - \mathbf{s}_k)/|\mathbf{r}_k - \mathbf{s}_k|$$

$r_k$  is the *a priori* mobile unit **10** location vector, obtained by forward propagation as described above.  $s_k$  is the position vector of satellite **70** at time  $t_k$ , obtained by GPS processor **20** using the satellite ephemeris and time model of satellite **70**.

A measurement obtained from a signal from a BTS **44** is a scalar (a one-  
5 component vector),  $z=\rho$ . The matrix  $H_k$  that relates the state vector  $e$  to this measurement vector at time  $t_k$  is:

$$H_k = \begin{pmatrix} u_x & u_y & u_z & 0 & 0 & 0 & 1 & 0 \end{pmatrix}$$

where the vector  $u=(u_x, u_y, u_z)$  is a unit vector from mobile unit **10** to BTS **44**. Note that the positions of BTSs **44** are fixed and known.

10 In both cases, an estimate of  $e$  is obtained by multiplying the measurement by the Kalman gain matrix  $K_k$ :

$$K_k = P_k H_k^T (H_k P_k H_k^T + R_k)^{-1}$$

$$e = K_k z$$

where the measurement noise covariance matrix  $R_k$  is:

$$15 \quad R_k = \begin{pmatrix} \sigma_r^2 & 0 \\ 0 & \sigma_{dz/dt}^2 \end{pmatrix}$$

for a satellite measurement and

$$R_k = \begin{pmatrix} \sigma_r^2 \end{pmatrix}$$

for a BTS measurement. The diagonal elements of  $\sigma_k$  are the variances of the corresponding pseudorange and pseudorange rate of change measurement errors. A

20 new *a posteriori* estimate of  $P$  is obtained as follows:

$$P_k^* = (I - K_k H_k) P_k$$

where  $I$  is the unit matrix. Finally, the *a posteriori* navigation vector  $\mathbf{n}_k^+$  is obtained by adding the state vector  $\mathbf{e}$  to the *a priori* navigation vector  $\mathbf{n}_k$ :  $x_k^+ = x_k + \delta x$ ,  $y_k^+ = y_k + \delta y$ ,  $z_k^+ = z_k + \delta z$ ,  $v_{xk}^+ = v_{xk} + \delta v_x$ ,  $v_{yk}^+ = v_{yk} + \delta v_y$ ,  $v_{zk}^+ = v_{zk} + \delta v_z$ ,  $c_{bk}^+ = c_{bk} + \delta c_b$  and  $c_{dk}^+ = c_{dk} + \delta c_d$ .

Figure 4 is a flow chart of the overall processing effected by joint estimator **30** to track the motion of mobile unit **10**. The processing is initialized (block **100**) with an initial estimate of the navigation vector  $\mathbf{n}$ , and by setting the diagonal elements of the covariance matrix  $\mathbf{P}$  equal to initial values that reflect the initial standard deviations of the errors in  $\mathbf{r}$ ,  $\mathbf{v}$ ,  $c_b$  and  $c_d$ . How the initial estimate of  $\mathbf{n}$  is obtained is discussed below. In block **102**, joint estimator **30** determines whether a measurement vector  $\mathbf{z}$  has been obtained from GPS processor **20** or from cellular processor **28**. If a measurement has not been obtained, then joint estimator **30** propagates (block **104**) the navigation vector  $\mathbf{n}$  forward in time, as described above. If a measurement has been obtained, then joint estimator uses the Kalman filter described above (block **106**) to obtain an *a posteriori* estimate of the navigation vector  $\mathbf{n}$ . Note that this updating of  $\mathbf{n}$  is based on a single new measurement, in contrast to the use of a Kalman filter in prior art GPS navigation, in which at least four independent simultaneous measurements are needed. This *a posteriori* estimate is reported (block **108**), and joint estimator **30** again checks for a new measurement vector  $\mathbf{z}$  from GPS processor **20** or from cellular processor **28**.

There are several methods by which the initial estimate of the navigation vector  $\mathbf{n}$  may be obtained. Most simply, a static estimate is used as the initial estimate of  $\mathbf{r}$  in block **100**, and  $\mathbf{v}$  is initially set to zero. By "static" estimation is meant that the estimation is performed under the assumption that mobile unit **10** did not move while the measurements were made and that the range offset  $c_b$  also was stationary while the

measurements were made. The simplest “static” estimate of  $\mathbf{r}$  is the location of a predefined reference point, such as the BTS **44** nearest to mobile unit **10**.  $c_b$  and  $c_d$  also are initialized to zero. This estimate is suitable for use in block **100** of Figure 4 if the first four measurements are satellite measurements, because the error in using the location of a terrestrial reference point is much smaller than the distances to the satellites. Triangulation is used to combine initial measurements from satellites with initial measurements from BTSs **44**, with the proviso that the equation used with a measurement from a BTSs **44** is the full (quadratic) triangulation equation

$$|\mathbf{s}-\mathbf{r}|^2 = \rho - c_b$$

rather than the linearized equation because BTSs **44** are too close to mobile unit **10** to justify linearization of this equation. Mathematical techniques for solving simultaneous quadratic and linear equations are well-known and need not be elaborated here.

Once the tracking of mobile unit **10** has commenced, the processing described above may be made both more accurate and more efficient. For example, the Doppler frequency shifts of the signals  $G_j(t)$  may be estimated more accurately. The total Doppler frequency shift is the sum of three terms: the Doppler frequency shift due to the motion of satellite  $j$ , the Doppler frequency shift due to the motion of mobile unit **10**, and a contribution from the drift of clock **22**. The first term can be computed from the ephemeris and time model of satellite  $j$ . Mobile unit **10** moves sufficiently slowly that the second term is negligible. The main initial uncertainty in the total Doppler frequency shift lies in the third term; but this term may be inferred from the range drift component  $c_d$  of the navigation vector  $\mathbf{n}$ . It then is possible to obtain an estimate of  $\dot{h}$ , the integer part of the Doppler frequency shift of  $G_j(t)$ , that is much more accurate

than the initial estimate described above, for use in the Doppler compensation factors for matrix  $M_{IJ}$ . Furthermore, the columns of  $M_j$  that are most likely to contain the peak corresponding to  $\rho_j$  and  $d\rho_j/dt$  can be predicted to within 10 to 20 Hz, as opposed to the 1 KHz range of the full matrix  $M_f$ , so that only these columns of matrix  $M_{IJ}$  need to be produced by the discrete orthogonal transform of matrix  $M_{Ij}$ .

Alternatively, having computed a pseudorange and a rate of change thereof for one satellite 70, mobile unit estimates the clock drift term of the total Doppler frequency shift by subtracting the computed Doppler frequency shift of the nearest BTS 44 from the apparent Doppler frequency shift that corresponds to the rate of change of the pseudorange to this one satellite 70. This estimated clock drift term is added to the known Doppler frequency shifts due to the motion of the other satellites 70 to provide estimates of the total Doppler frequency shifts of the other satellites 70.

While the invention has been described with respect to a limited number of embodiments, it will be appreciated that many variations, modifications and other applications of the invention may be made.

## WHAT IS CLAIMED IS:

1. A method for determining a pseudorange, and a rate of change thereof, to a beacon that transmits a signal including a plurality of blocks, each block including a plurality of frames of a pseudonoise sequence, each block being multiplied by a bit of a data sequence, the signal being shifted relative to a nominal frequency by a frequency shift, the method comprising the steps of:
- (a) receiving the signal;
  - (b) digitizing said received signal, thereby producing a digitized signal including a plurality of bits;
  - (c) arranging said digitized signal as columns of an input matrix that includes a plurality of rows, each said column including consecutive said bits of said digitized signal that correspond to an integral number of frames of the pseudonoise sequence; and
  - (d) performing a discrete orthogonal transform on each said row of said input matrix, thereby producing a transformed matrix.
2. The method of claim 1, wherein said integral number is 1.
3. The method of claim 1, wherein said discrete orthogonal transform is a discrete Fourier transform.
4. The method of claim 1, wherein said transformed matrix includes a plurality of elements, the method further comprising the step of:

- (e) multiplying each said element of said transformed matrix by a respective Doppler compensation factor.

5. The method of claim 4, further comprising the step of:

- 5 (f) providing an estimate of the frequency shift;  
said Doppler compensation factors being based on said estimate of the frequency shift.

6. The method of claim 5, wherein said transformed matrix has fewer said columns than said input matrix, said estimate of the frequency shift being used to  
10 determine which said columns of said transformed matrix are produced by said discrete orthogonal transform.

7. The method of claim 4, further comprising the step of:

- (f) convolving each said column of said transformed matrix with the  
15 pseudonoise sequence.

8. The method of claim 7, further comprising the step of:

- (g) identifying at least one peak in said transformed matrix, each said at  
least one peak having a row coordinate and a column coordinate.

20

9. The method of claim 8, further comprising the step of:

- (h) selecting, from among said at least one peak, a most likely peak, said row coordinate of said most likely peak then corresponding to the

pseudorange, and said column coordinate of said most likely peak then corresponding to the rate of change of the pseudorange.

10. The method of claim 8, wherein said receiving, said digitizing, said  
5 arranging, said performing, said multiplying, said convolving and said identifying are effected on a plurality of instances of the transmitted signal, the method further comprising the step of:

- (h) inferring the pseudorange and the rate of change of the pseudorange from said row coordinates and said column coordinates.

10

11. The method of claim 10, wherein said inferring is effected using a track-before-detect identification scheme.

12. The method of claim 7, wherein said receiving, said digitizing, said  
15 arranging, said performing, said multiplying and said convolving are effected on a plurality of instances of the transmitted signal to produce a corresponding plurality of said transformed matrices, the method further comprising the steps of:

- (h) integrating said transformed matrices non-coherently to produce a sum matrix; and
- 20 (i) identifying at least one peak in said sum matrix, each said at least one peak having a row coordinate and a column coordinate.



13. The method of claim 12, further comprising the step of:

- (j) selecting, from among said at least one peak, a most likely peak, said row coordinate of said most likely peak then corresponding to the pseudorange, and said column coordinate of said most likely peak then corresponding to the rate of change of the pseudorange.

14. The method of claim 4, further comprising the step of:

- (g) providing a plurality of estimates of the frequency shift, thereby producing, for each said element of said transformed matrix, a corresponding plurality of said respective Doppler compensation factors that are based on said estimates of the frequency shift;

and wherein said multiplying is effected separately for each said estimate of the frequency shift, thereby producing a corresponding plurality of Doppler-compensated matrices.

15. The method of claim 1, further comprising the step of:

- (e) multiplying groups of said bits of said digitized signal, that correspond to the blocks of the transmitted signal, by respective bits of the data sequence, prior to said arranging of said digitized signal as columns of said input matrix.

16. A method for determining a pseudorange, and a rate of change thereof, to each of a plurality of beacons, each beacon transmitting a respective signal including a plurality of blocks, each block including a plurality of frames of a

pseudonoise sequence, each block being multiplied by a bit of a data sequence, the pseudonoise sequences and the data sequences being uniquely associated with respective satellites, all the pseudonoise sequences being of equal length, the pseudonoise sequences being mutually orthogonal, the method comprising the steps

5 of:

- (a) receiving the transmitted signals collectively as a received signal;
- (b) digitizing said received signal, thereby producing a digitized signal including a plurality of bits; and
- (c) for each beacon, multiplying groups of said bits of said digitized signal, that correspond to the blocks of the signal transmitted by said each beacon, by respective bits of the data sequence of said each beacon.

10

17. The method of claim 16, further comprising the steps of:

- (d) for each beacon, arranging said digitized signal as columns of a corresponding instance of a first matrix that includes a plurality of rows, each said column including consecutive said bits of said digitized signal that correspond to an integral number of frames of the pseudonoise sequences; and

15

- (e) for each beacon, performing a discrete orthogonal transform on each said row of said corresponding instance of said first matrix.

20

18. The method of claim 17, wherein each said instance of said first matrix includes a plurality of elements, the method further comprising the steps of, for each beacon:

- (f) multiplying each said element of said corresponding instance of said first matrix by a respective Doppler compensation factor; and
- (g) convolving each said column of said corresponding instance of said first matrix with said pseudonoise sequence of said each beacon.

5

19. The method of claim 18, further comprising the step of:

- (h) for each beacon, identifying at least one peak in said corresponding instance of said first matrix, each said at least one peak having a row coordinate and a column coordinate.

10

20. The method of claim 19, further comprising the step of:

- (i) for each beacon, selecting, from among said at least one peak, a most likely peak, said row coordinate of said most likely peak then corresponding to the pseudorange of said each beacon, and said column coordinate of said most likely peak then corresponding to the rate of change of the pseudorange of said each beacon.

15

21. The method of claim 20, wherein said receiving, said digitizing, said multiplying by respective bits of the data sequence, said arranging, said performing, said multiplying of said elements by said Doppler compensation factors, said convolving and said identifying are effected on a plurality of instances of the transmitted signal, the method further comprising the step of:

- (j) for each beacon, inferring the pseudorange of said each beacon and the rate of change of the pseudorange of said each beacon from said row

coordinates and said column coordinates corresponding to said each beacon.

22. The method of claim 18, wherein said receiving, said digitizing, said multiplying by respective bits of the data sequence, said arranging, said performing, said multiplying of said elements by said Doppler compensation factors and said convolving are effected on a plurality of instances of the transmitted signal to produce, for each beacon, a corresponding plurality of said second matrices, the method further comprising the steps of: for each beacon:

- 10 (h) integrating said second matrices non-coherently to produce a sum matrix; and
- (i) identifying at least one peak in said sum matrix, each said at least one peak having a row coordinate and a column coordinate.

15 23. The method of claim 22, further comprising the step of: for each beacon

- (j) selecting, from among said at least one peak, a most likely peak, said row coordinate of said most likely peak then corresponding to the pseudorange, and said column coordinate of said most likely peak then corresponding to the rate of change of the pseudorange.

24. The method of claim 16, wherein said respective bits of said data sequences are common to all the beacons.

25. A receiver for receiving a signal transmitted by a beacon, the signal including a plurality of frames of a pseudonoise sequence, comprising:

- (a) an antenna for receiving the transmitted signal;
- (b) a mechanism for digitizing the received signal to produce a digitized signal including a plurality of bits;
- (c) a memory for storing the digitized signal as columns of a matrix that includes a plurality of rows, each said column including consecutive said bits of said digitized signal that correspond to an integral number of frames of the pseudonoise sequence; and
- (d) a processor for performing a discrete orthogonal transform on each of said rows of said matrix.

26. A locator system for locating a mobile unit, comprising:

- (a) at least one beacon having a respective pseudonoise sequence and a respective data sequence, each said data sequence including a plurality of bits, each said at least one beacon operative to transmit a respective transmitted signal, each said respective transmitted signal including a plurality of blocks, each said block including a plurality of frames of said respective pseudonoise sequence, each said block being multiplied by a bit of said respective data sequence,

- (b) a reference unit including:
  - (i) a reference unit receiver for:
    - (A) receiving said at least one transmitted signal as a reference unit received signal, and

- (B) recovering said at least one data sequence from said  
respective reference unit received signal, and
  - (ii) a transmitter for transmitting said at least one data sequence to  
the at least one mobile unit; and
  - 5 (c) in the mobile unit:
    - (i) a first mobile unit receiver for receiving said at least one data  
sequence, and
    - (ii) a second mobile unit receiver including:
      - (A) an antenna for receiving the at least one transmitted  
10 signal collectively as a mobile unit received signal,
      - (B) a mechanism for digitizing the mobile unit received  
signal to produce a digitized signal including a plurality  
of bits,
      - (C) a memory for storing, for each of the at least one  
15 beacon, an instance of said digitized signal, and
      - (D) a processor for multiplying each of said at least one  
instance of said digitized signal by said bits of said  
respective data sequence.

20           27. The locator system of claim 26, wherein each said at least one beacon  
is operative to move on a known respective trajectory while transmitting said  
respective transmitted signal.

28. The locator system of claim 26, wherein said processor is operative, for each of the at least one beacon, to arrange said corresponding instance of said digitized signal in said memory as columns of a matrix that includes a plurality of rows, each said column including consecutive said bits of said corresponding instance  
 5 that correspond to an integral number of frames of said respective pseudonoise sequence, and to perform a discrete orthogonal transform on each of said rows of said matrix.

29. A method for determining a location of a receiver, comprising the steps  
 10 of:
- (a) providing a plurality of beacons having respective pseudonoise sequences, all said pseudonoise sequences being of equal length;
  - (b) transmitting, by each said beacon, a respective signal including a plurality of frames of said respective pseudonoise sequence;
  - 15 (c) receiving said transmitted signals collectively as a received signal, by the receiver;
  - (d) inferring, for each beacon, a pseudorange and a rate of change of said pseudorange; and
  - (e) inferring the location of the receiver from said pseudoranges and from  
 20 said rates of change of said pseudoranges.

30. The method of claim 29, wherein said inferring of said pseudoranges and said rates of change is effected by steps including:

- (i) digitizing said received signal, thereby producing a digitized signal including a plurality of bits;
- (ii) for each said beacon, arranging said digitized signal as columns of a matrix that includes a plurality of rows, each said column including consecutive said bits of said digitized signal that correspond to an integral number of said frames of said pseudonoise sequences; and
- (iii) for each said beacon, performing a discrete orthogonal transform on each said row of said matrix.

31. A method for determining a pseudorange, and a rate of change thereof, to a beacon that transmits a signal including a plurality of blocks, each block including a plurality of frames of a pseudonoise sequence, each block being multiplied by a bit of a data sequence, the signal being shifted relative to a nominal frequency by a Doppler frequency shift, the method comprising the steps of:

- (a) receiving the signal;
- (b) digitizing said received signal, thereby producing a digitized signal including a plurality of bits; and
- (c) applying a matched filter algorithm to said digitized signal to extract therefrom the pseudorange and the rate of change of the pseudorange, said matched filter algorithm including:
  - (i) demodulating said digitized signal relative to the data sequence.



32. The method of claim 31, further including the step of:

- (d) aligning the bits of the data sequence with said digitized signal, prior to said demodulation.

5 33. The method of claim 32, wherein said aligning is within about one millisecond.

34. The method of claim 31, wherein said matched filter algorithm further includes:

- 10 (ii) arranging said digitized signal as columns of a matrix that includes a plurality of rows, each said column including consecutive said bits of said digitized signal that correspond to an integral number of frames of the pseudonoise sequence; and
- (iii) performing a discrete orthogonal transform on each said row.

15 35. The method of claim 34, wherein said matrix includes a plurality of elements, and wherein said matched filter algorithm further includes:

- (iv) multiplying each said element by a respective Doppler compensation factor.

20

36. The method of claim 35, wherein said matched filter algorithm further includes:

- (v) convolving each said column with the pseudonoise sequence; and

- (vi) identifying at least one peak in said matrix, each said at least one peak having a row coordinate and a column coordinate.

37. The method of claim 36, wherein said receiving, said digitizing, said  
 5 arranging, said performing, said multiplying, said convolving and said identifying are effected on a plurality of instances of the transmitted signal, and wherein said matched filter algorithm further includes:

- (vii) inferring the pseudorange and the rate of change of the pseudorange from said row coordinates and said column coordinates.

10 38. The method of claim 36, wherein said receiving, said digitizing, said arranging, said performing, said multiplying and said convolving are effected on a plurality of instances of the transmitted signal to produce a corresponding plurality of said matrices, and wherein said matched filter algorithm further includes:

- 15 (viii) integrating said matrices non-coherently to produce a sum matrix;
- (ix) identifying at least one peak in said sum matrix, each said at least one peak having a row coordinate and a column coordinate; and
- (x) selecting, from among said at least one peak, a most likely peak, said  
 20 row coordinate of said most likely peak then corresponding to the pseudorange, and said column coordinate of said most likely peak then corresponding to the rate of change of the pseudorange.

## ABSTRACT OF THE DISCLOSURE

A method and system for finding the position of a mobile unit with respect to the satellites of a satellite network such as the Global Positioning System and with respect to the base stations of a wireless communications network. Each satellite transmits a signal that consists of a series of frames of a pseudonoise sequence. The frames of a signal received from the satellite network by the mobile unit are arranged as columns of a matrix and are processed coherently to provide estimated pseudoranges and estimated rates of change of pseudoranges for in view satellites. The coherent processing includes performing an orthogonal transform on the rows of the matrix, multiplying the elements of the matrix by Doppler compensation factors, and then, for each satellite in view, convolving the columns of the matrix with the pseudonoise sequence of that satellite. Other pseudoranges are inferred from synchronization burst sequences received by the mobile unit from one or more base stations. If the base stations are not mutually synchronized, reference units are provided to determine the time offsets between the base station clocks and the satellite network clock and to transmit these time offsets to the mobile unit, so that all the pseudoranges include the same bias relative to the satellite network clock. The pseudoranges are processed and reconciled by a dynamic estimator that includes a linear optimal estimator such as a Kalman filter, to produce successive estimates of the location of the mobile unit as the mobile unit moves.



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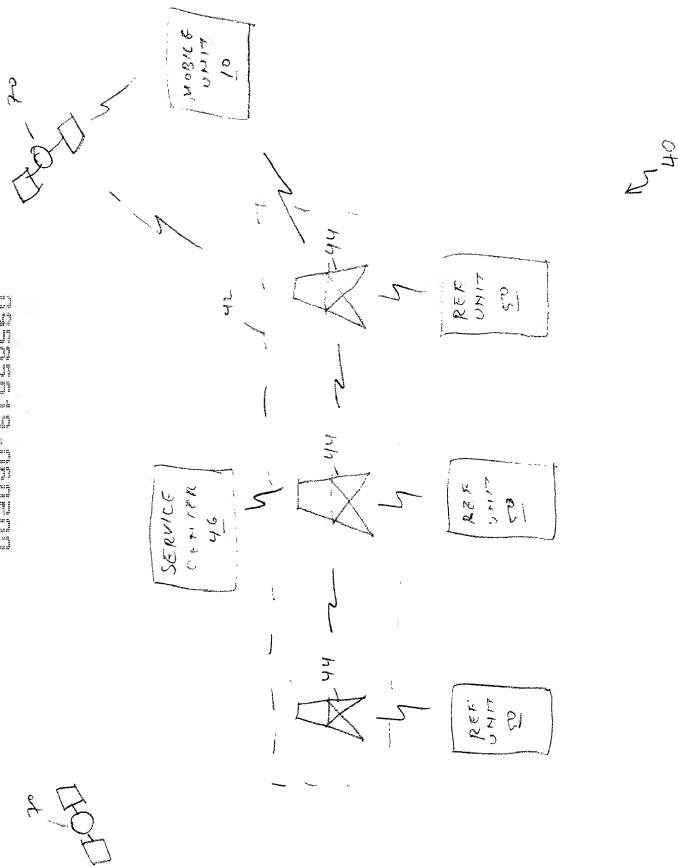


FIG. 2

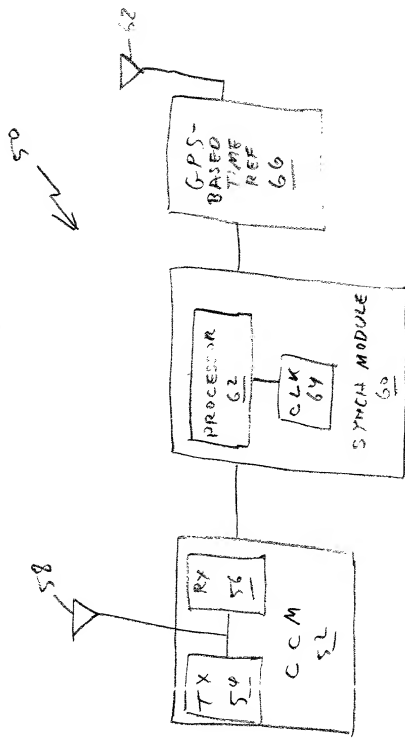


FIG 3

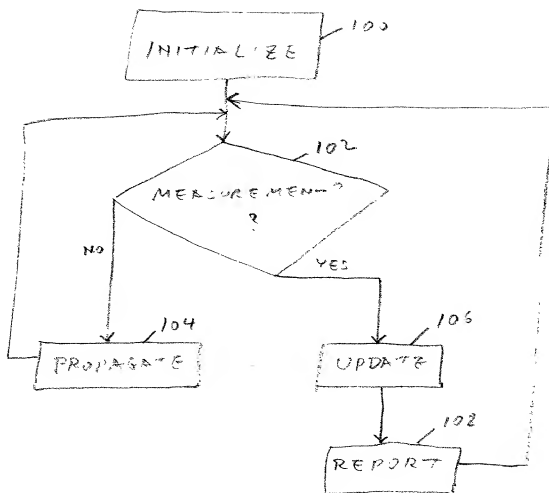


FIG. 4

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Combined Declaration For Patent Application and Power of Attorney

As a below named inventor, I hereby declare that:

My residence, post office address and citizenship are as stated below next to my name;

I believe I am the original, first and sole inventor (if only one name is listed below) or an original, first and joint inventor (if plural names are listed below) of the subject matter which is claimed and for which a patent is sought on the invention entitled COHERENT PROCESSING OF SATELLITE SIGNALS TO LOCATE A MOBILE UNIT, the specification of which

(check one) ☒ is attached hereto.

☐ was filed on \_\_\_\_\_ as Application Serial No. \_\_\_\_\_ and was amended on \_\_\_\_\_. I hereby state that I have reviewed and understand the contents of the above identified specification, including the claims, as amended by any amendment referred to above.

I acknowledge the duty to disclose information which is material to the patentability of this application in accordance with Title 37, Code of Federal Regulations, § 1.56(a).

I hereby claim foreign priority benefits under Title 35, United States Code, § 119 of any foreign application(s) for patent or inventor's certificate listed below and have also identified below any foreign application for patent or inventor's certificate having filing date before that of the application on which priority is claimed:

Prior Foreign Application(s)

NONE

(number)

(Country)

(Day, Month, Year Filed)

Priority Claimed

☐ ☐

Yes No

(number)

(Country)

(Day, Month, Year Filed)

☐ ☐

Yes No

(number)

(Country)

(Day, Month, Year Filed)

☐ ☐

Yes No

I hereby claim the benefit under Title 35, United States Code, § 120 of any United States Application(s) listed below and, insofar as the subject matter of each of the claims of this application is not disclosed in the prior United States application in the manner provided by the first paragraph of Title 35, United States code, § 112, I acknowledge the duty to disclose material information as defined in Title 37, Code of Federal Regulations, § 1.56(a) which occurred between the filing date of the prior application and the national or PCT international filing date of this application:

NONE

(Application Serial No.)

(Filing Date)

Status

(patented, pending, abandoned)

(Application Serial No.)

(Filing Date)

Status

(patented, pending, abandoned)

I hereby appoint the following attorneys, with full power of substitution, association, and revocation, to prosecute this application and to transact all business in the Patent and Trademark Office connected therewith.

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004/007

Attorney Docket: 1759/2

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## Continuation of Combined Declaration For Patent Application and Power of Attorney

I hereby further declare that all statements made herein of my own knowledge are true and that all statements made on information and belief are believed to be true; and further that these statements were made with the knowledge that willful false statements and the like so made are punishable by fine or imprisonment, or both, under Section 1001 of Title 18 of the United States Code and that such willful false statement may jeopardize the validity of the application of any patent issued thereon.

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FULL NAME OF FOURTH INVENTOR	INVENTOR'S SIGNATURE	DATE
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POST OFFICE ADDRESS		

FULL NAME OF FIFTH INVENTOR	INVENTOR'S SIGNATURE	DATE
RESIDENCE	CITIZENSHIP ISRAELI	
POST OFFICE ADDRESS		

FULL NAME OF SIXTH INVENTOR	INVENTOR'S SIGNATURE	DATE
RESIDENCE	CITIZENSHIP ISRAELI	
POST OFFICE ADDRESS		